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HelioSmart



Reimagined Solar Energy

Section 13337, Group 9

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POWERING THE NEW ENGINEER TO TRANSFORM THE FUTURE

Meet the HelioSmart Team



Adrien Arias



Jarett Cox



Olivia Dodge



Zariq George



Cristian Hooker



Abraham Sheikh

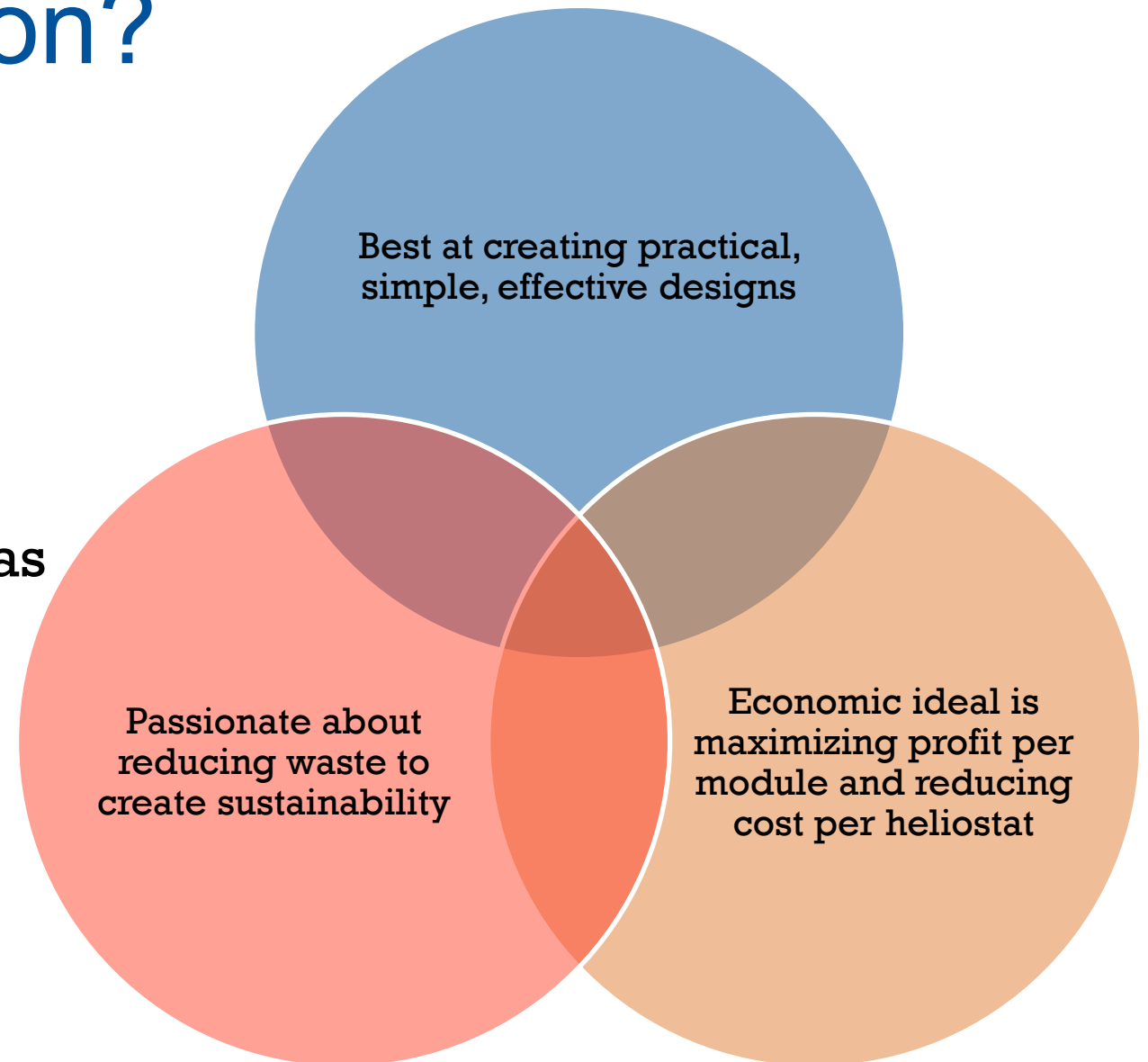


Noelle Turner

What drives our mission?

Emphasis on **efficiency and cost**

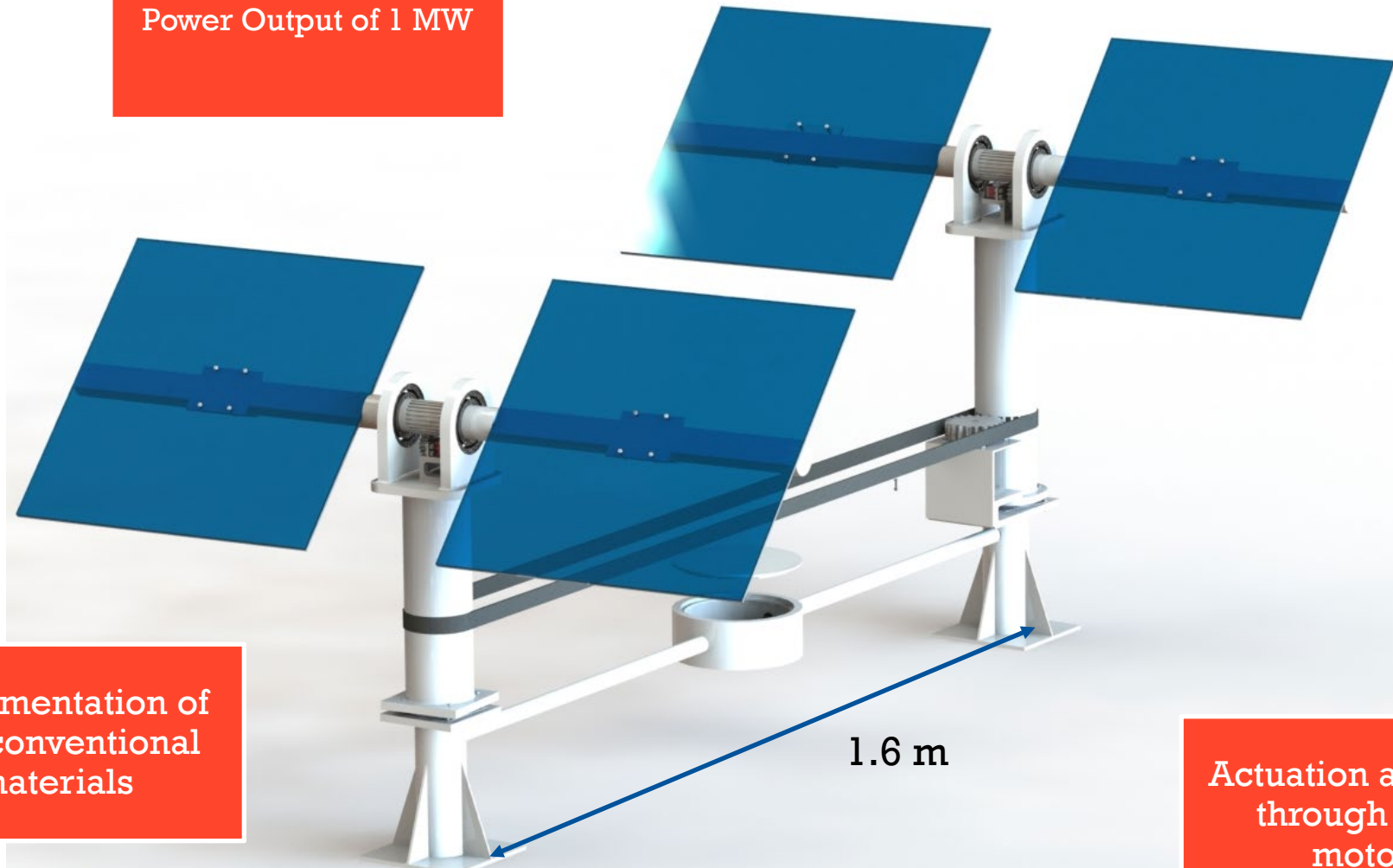
- Use low-cost novel materials
- Assure 20-year lifetime
- Handle desert conditions in Las Vegas
- Reduce waste



- Designing a heliostat module that is meant to be a part of a plant
- Located in Las Vegas, Nevada
- Reducing cost while optimizing a small and compact design
- Smaller heliostats not currently utilized commercially



Power Output of 1 MW



Implementation of
non-conventional
materials

1.6 m

Actuation achieved
through three
motors

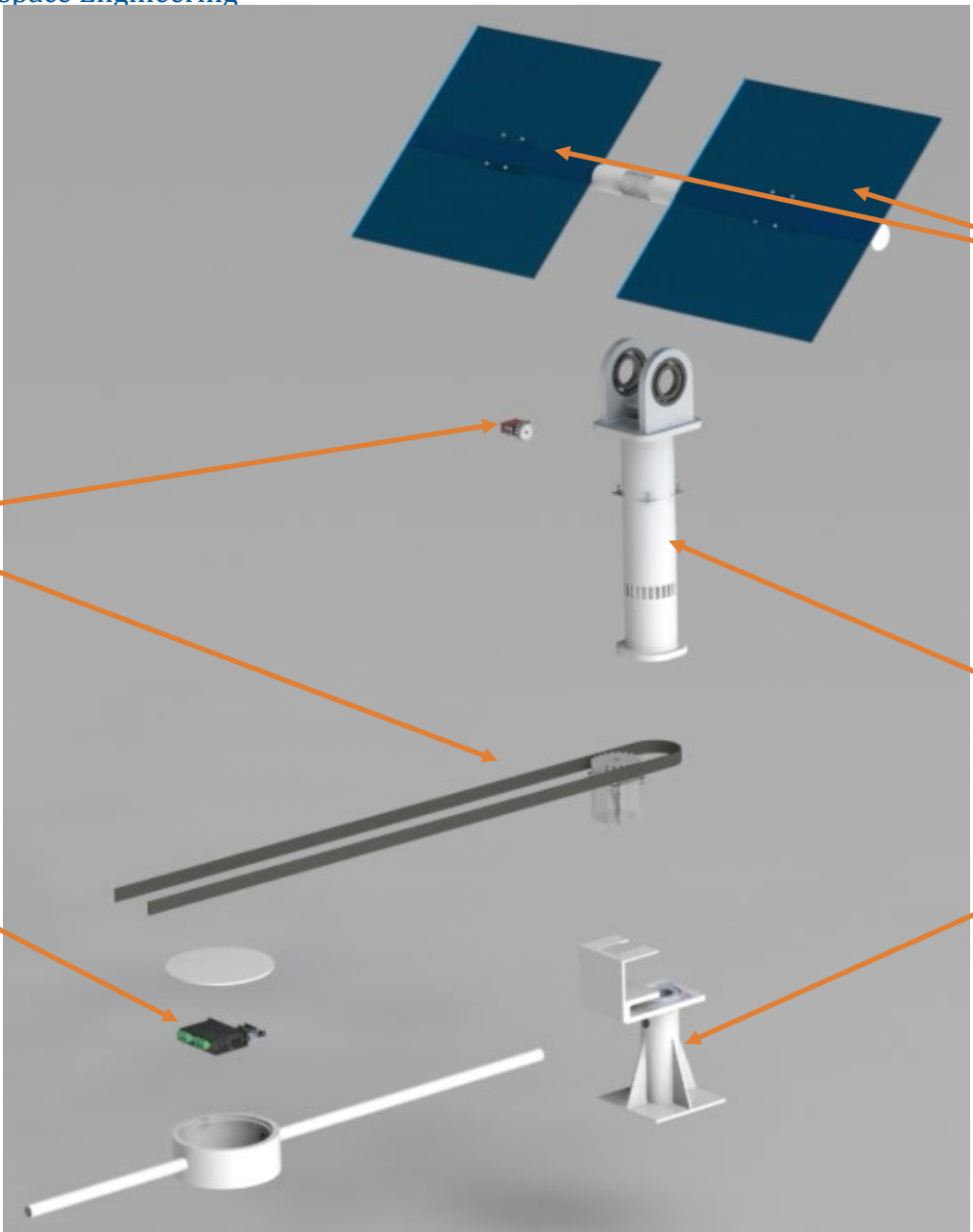
Key Features

Actuation
Subsystem

Controller
Feedback
Subsystem

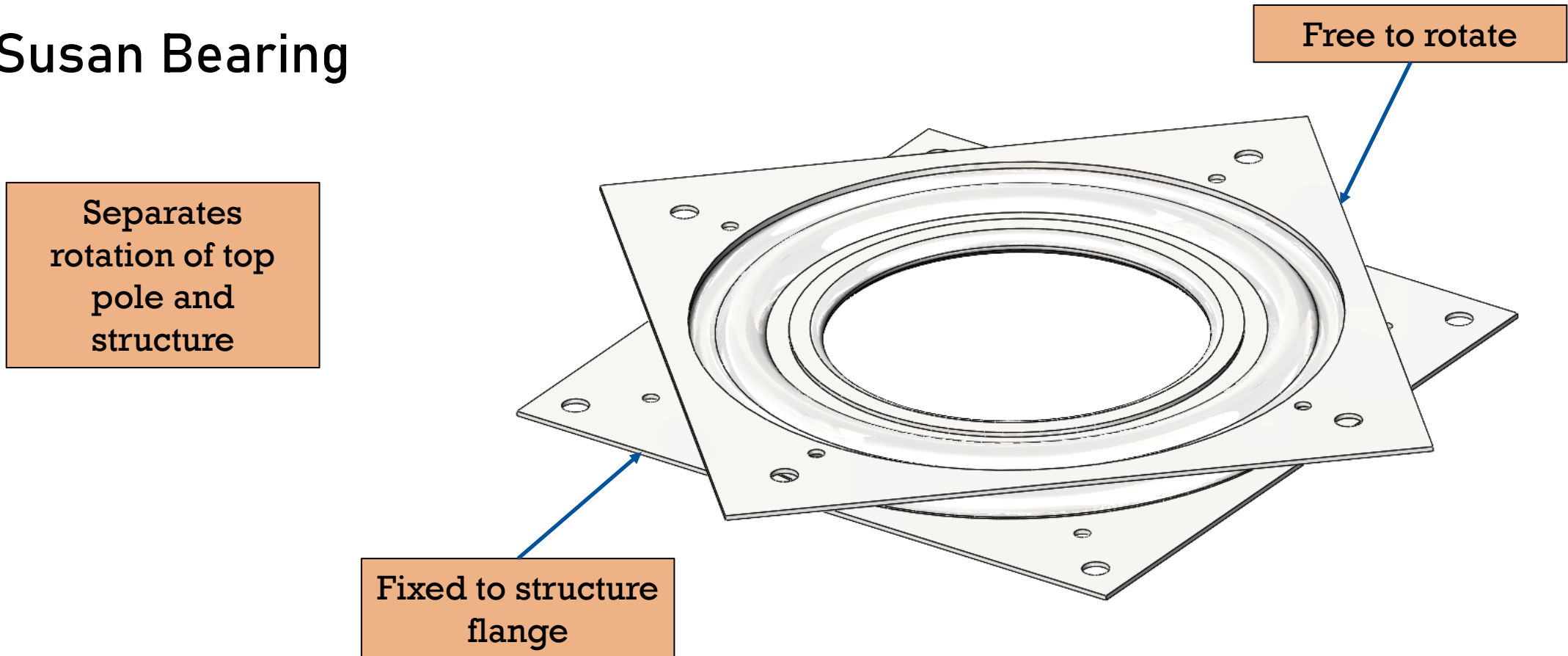
Reflective
Surface
Subsystem

Structure
Subsystem

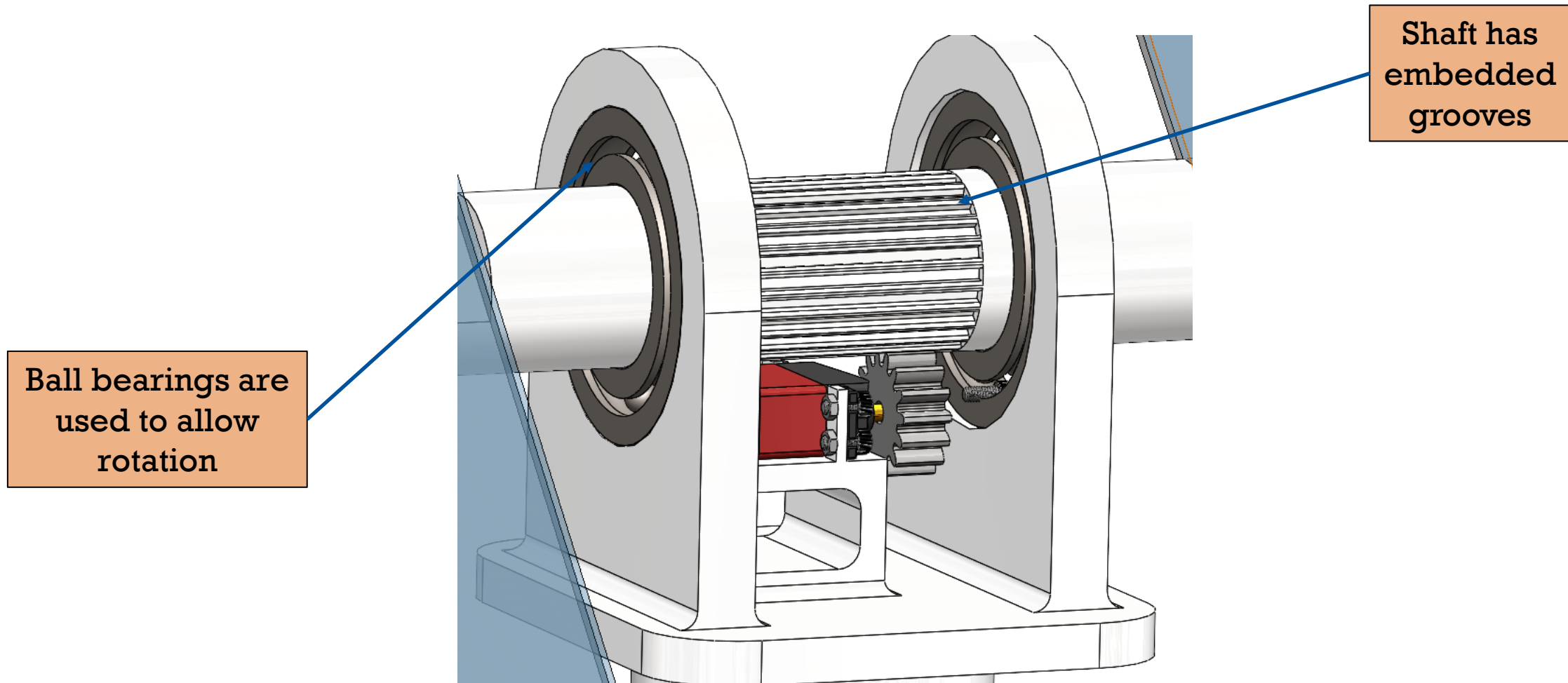


Unique Design Characteristics

Lazy-Susan Bearing



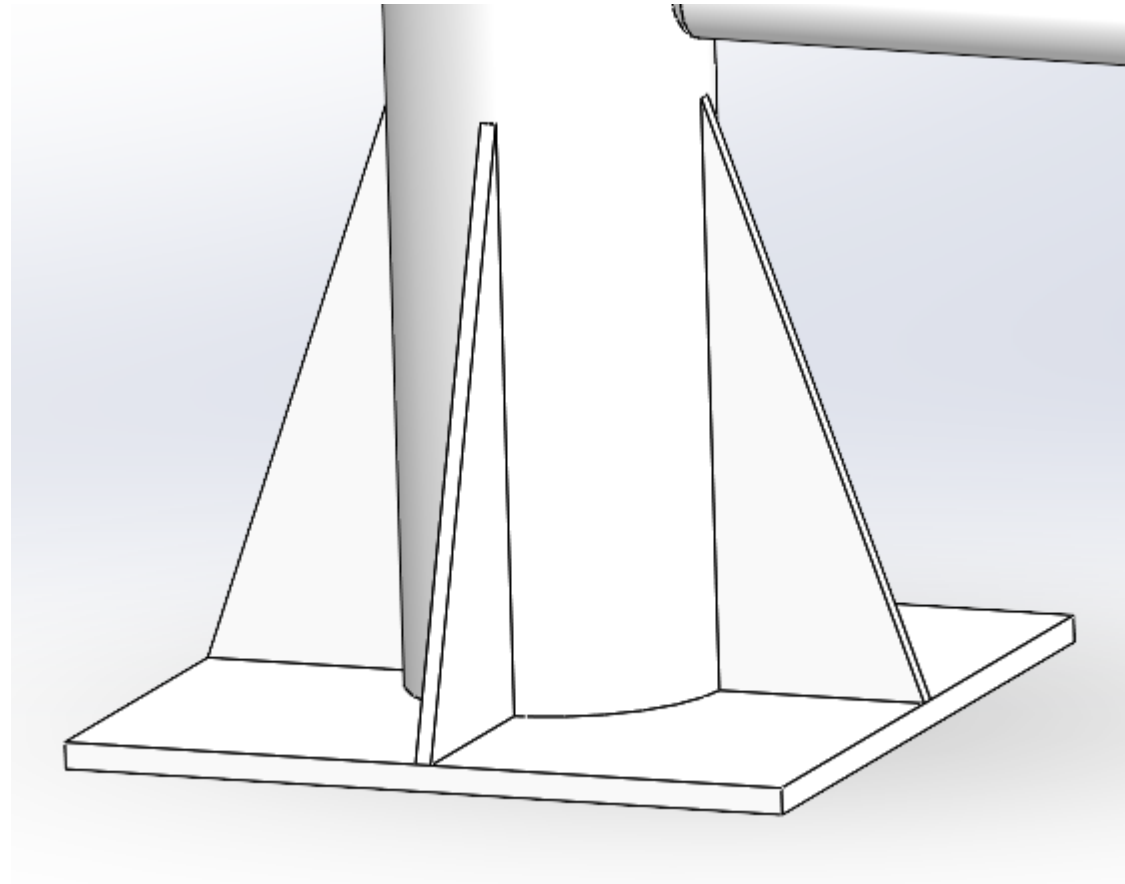
Gear to Pole Connection



Ground Structure

Anchors module

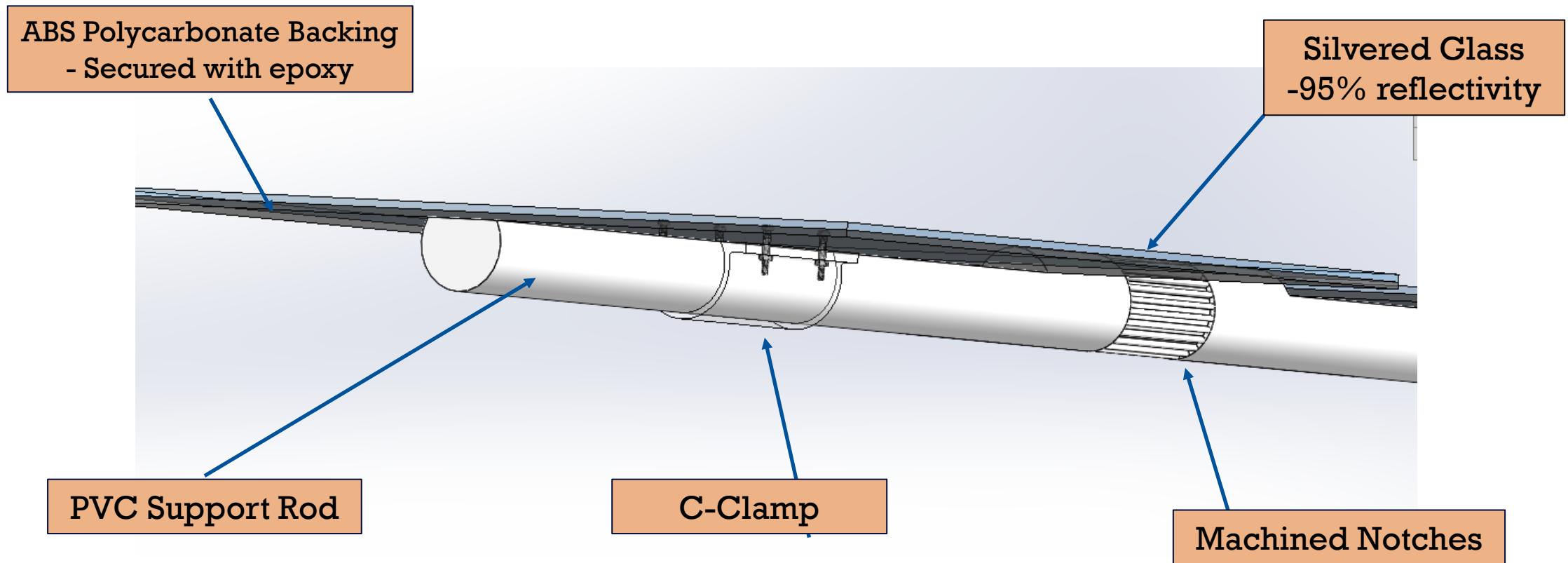
- underground
- flanges prevent rotation
- flat base prevents uplift



Reflective Surface Subsystem

Need	Metric	Solution
1	Number of innovations	0.25 m ² heliostat size
2	Total Collection area $\leq 1 \text{ m}^2$	4 square 0.25 m ² reflective surfaces
8	Overall cost $\leq \$100/\text{m}^2$	Low-cost plane mirrors
12	Withstand surface pressure up to 500 psi	Standard glass surface
14	Operational lifetime ≥ 20 years	Glass surface with ABS frame
15	$35^\circ \text{ F} \leq \text{Operating temperature} \leq 110^\circ \text{ F}$	Silver-backed glass mirror
16	$Q_{in} \geq 1.05 \text{ MW}$	1627 modules to generate power
17	Solar concentration ratio $\geq 1000 \text{ kW}/\text{m}^2$	Compact module for higher ratio
18	Total Integrated Scatter $\leq 5\%$	Smooth reflective surface with 1.65% TIS

Reflective Surface Design



Reflective Surface Subsystem Analyses

- Optical Efficiency:

$$\eta_{opt} = \rho * \eta_{cos} = (0.95)(0.77) = 0.71$$

- Solar Radiation Flux:

$$G_{bn} = \int_0^{30^\circ} (1000 \text{ W/m}^2) \cos(\theta) d\theta = 866 \text{ W/m}^2$$

- $0 \leq \theta \leq 30^\circ$ on December 21st, 2020

Reflective Surface Subsystem Analyses

- Useful Input Power (1 MW)

$$\dot{Q}_{use} = G_{bn} * \eta_{opt} * n * A_{module}$$

Where n is the number of heliostat modules.

$$1 \text{ MW} = (866 \text{ W/m}^2)(0.71)n(1\text{m}^2) = 1627 \text{ modules}$$

$$\dot{Q}_{use} = \dot{Q}_{in} * \eta_{opt} \Rightarrow \dot{Q}_{in} = \boxed{1.41 \text{ MW}}$$

Reflective Surface Subsystem Analyses

■ Solar Concentration Ratio

$$C_{geo} = \frac{\Sigma A_{\text{heliostat modules}}}{A_{\text{receiver}}} = \frac{n * A_{\text{module}}}{A_{\text{receiver}}} = \frac{1627(1 \text{ m}^2)}{1 \text{ m}^2} = 1627 \text{ suns}$$

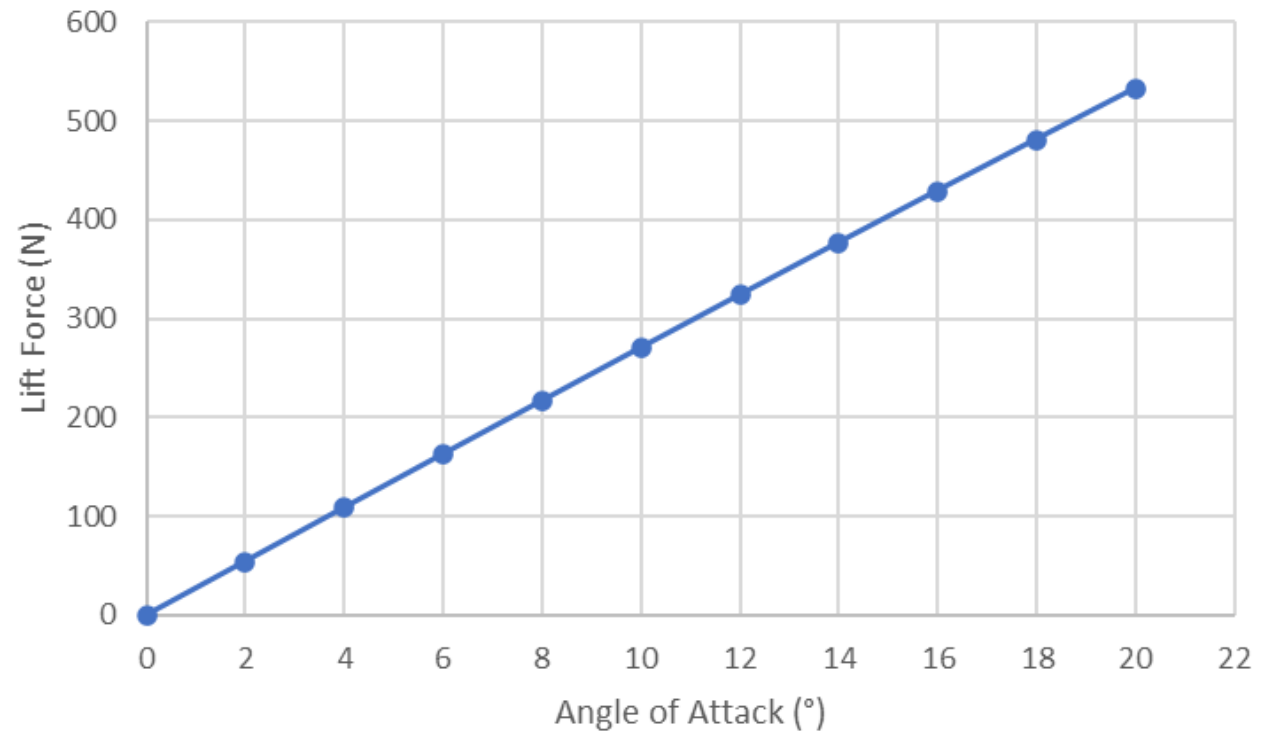
Reflective Surface Subsystem Analyses

- Lift force on a flat plate:

$$L = \frac{1}{2} \rho v^2 A C_L$$

- Coefficient of lift:

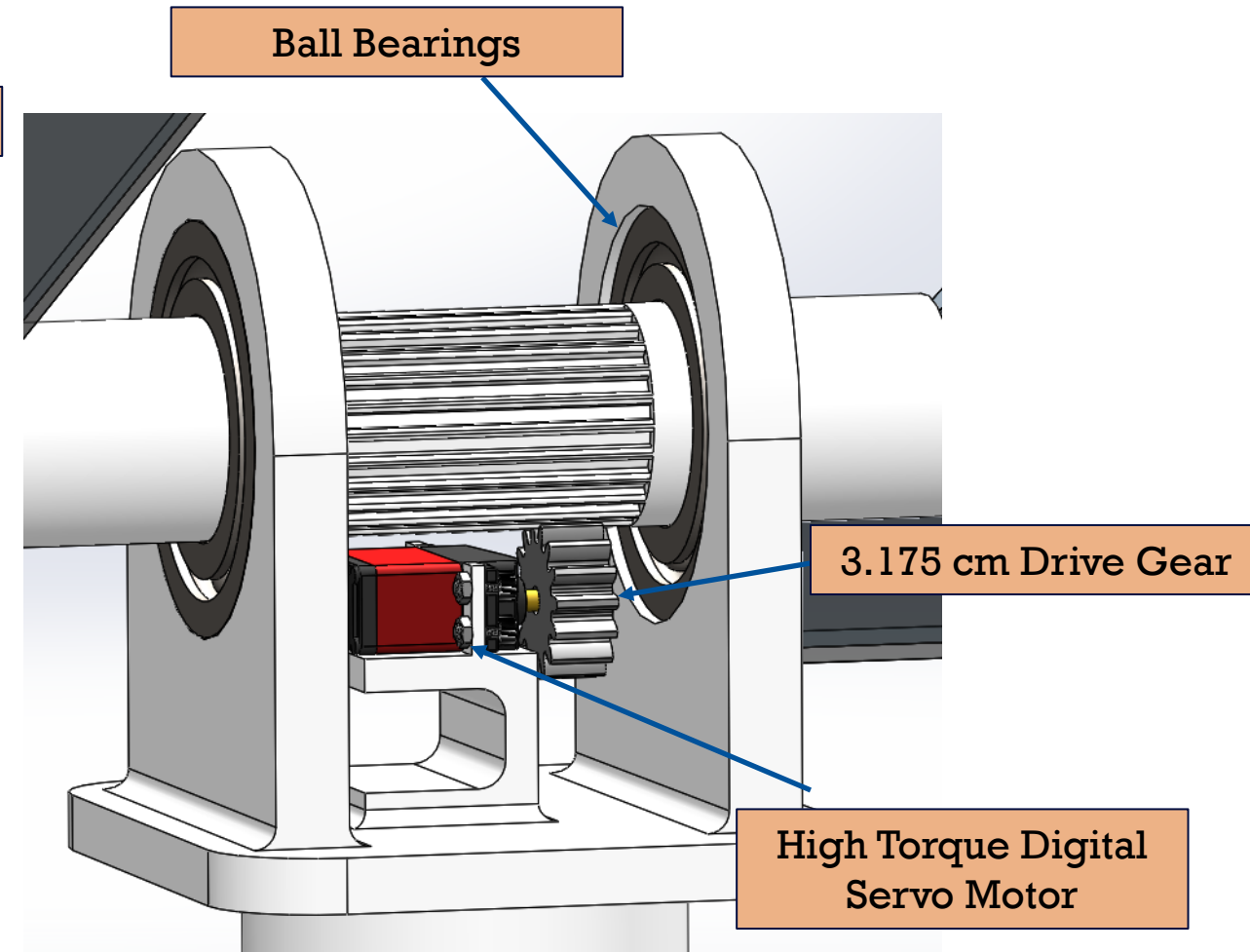
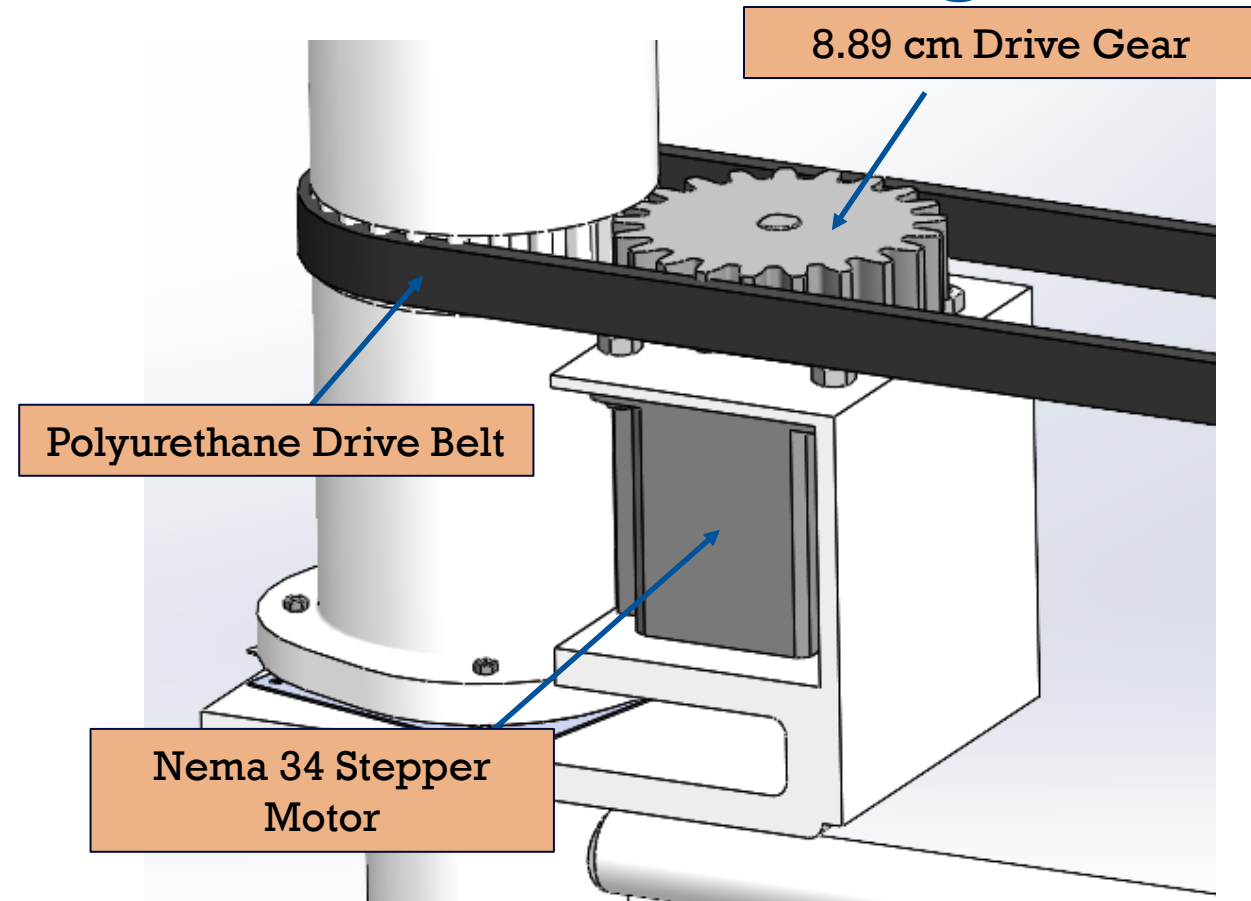
$$C_L = 2\pi \sin \alpha$$



Actuation Subsystem

Need	Metric	Solution
4	Tracking Error $\leq 0.5^\circ$	Stepper motor drivers for precision
7	Receiver elevation ≤ 100 m	Range of motion $>180^\circ$
8	Overall cost $\leq \$100/\text{m}^2$	Low-cost servo motors
11	Relative part cost $\geq \$0$	OTS parts preferred
14	Operational lifetime ≥ 20 years	20+ years with maintenance

Actuation Design



Actuation Subsystem Design Analyses

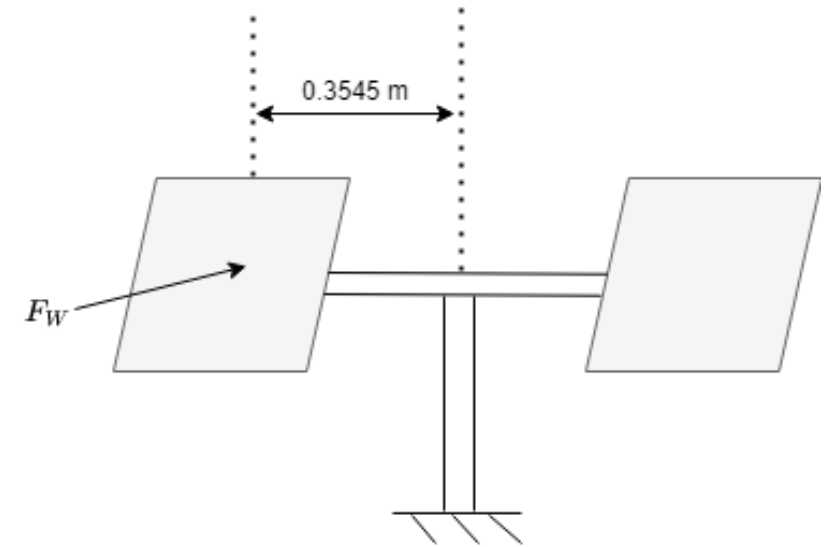
- Required torque for azimuthal axis

Wind force on each mirror:

$$F_w = \frac{1}{2} \rho v^2 A = \frac{1}{2} \left(1.2 \frac{\text{kg}}{\text{m}^3} \right) \left(4.47 \frac{\text{m}}{\text{s}} \right)^2 (0.25 \text{ m}^2) = 3.0 \text{ N}$$

Torque due to the wind:

$$T = F_w * r = 2(3 \text{ N})(0.3545 \text{ m}) = \boxed{2.125 \text{ N} \cdot \text{m}}$$



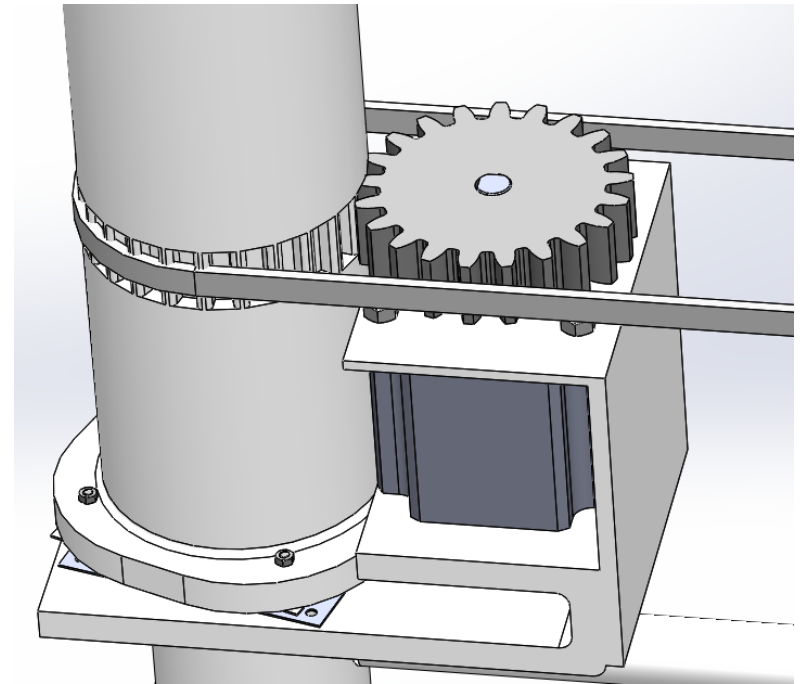
Actuation Subsystem Design Analyses

Motor Torque, $T_1 = 4.8 \text{ N} \cdot \text{m}$

Available torque due to gear ratio:

$$\frac{d_1}{d_2} = \frac{T_1}{T_2}, \text{ where } d_1 = 8.89 \text{ cm}; d_2 = 10.16 \text{ cm}$$

$$T_2 = 5.486 \text{ N} \cdot \text{m}$$



Actuation Subsystem Design Analyses

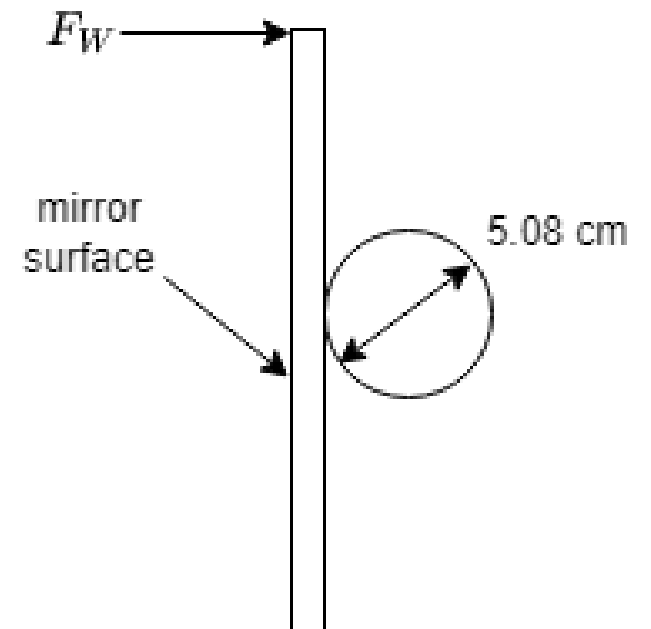
- Required torque for tilt axis (against wind force)

Wind force for average wind speed: $F_w = 2.997\text{ N}$

Torque: $T = F_w r = (2.997\text{ N})(0.25\text{ m}) = 0.749\text{ N} \cdot \text{m}$

Torque produced by drive gears:

$$\frac{d_1}{d_2} = \frac{T_1}{T_2} \rightarrow \frac{3.175\text{ cm}}{5.08\text{ cm}} = \frac{1.96\text{ N} \cdot \text{m}}{T_2} \rightarrow \boxed{3.14\text{ N} \cdot \text{m}}$$



Actuation Subsystem Design Analyses

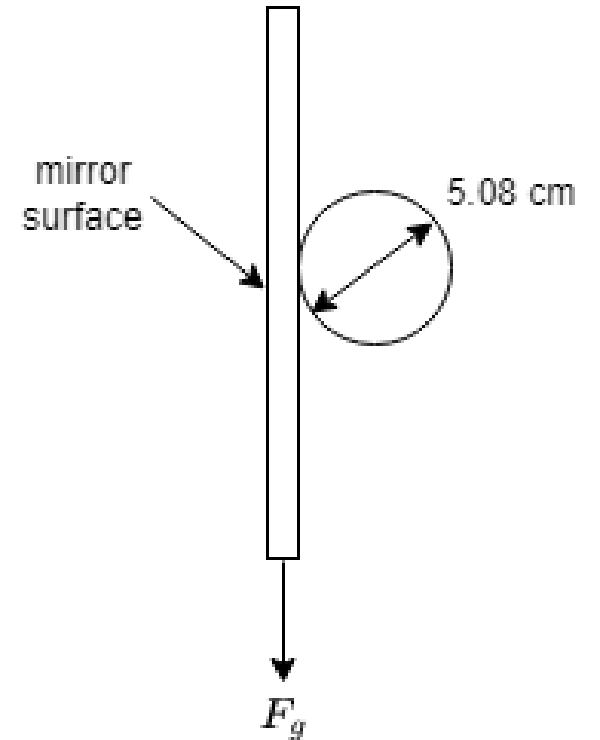
- Required torque for tilt axis (against gravity)

Weight of two mirrors: 6.95 kg $\rightarrow F_g = 68.18$ N

Torque: $T = F_w r = (68.18 \text{ N})(0.0254 \text{ m}) = 1.73 \text{ N} \cdot \text{m}$

Torque produced by drive gears: $T_2 = 3.14 \text{ N} \cdot \text{m}$

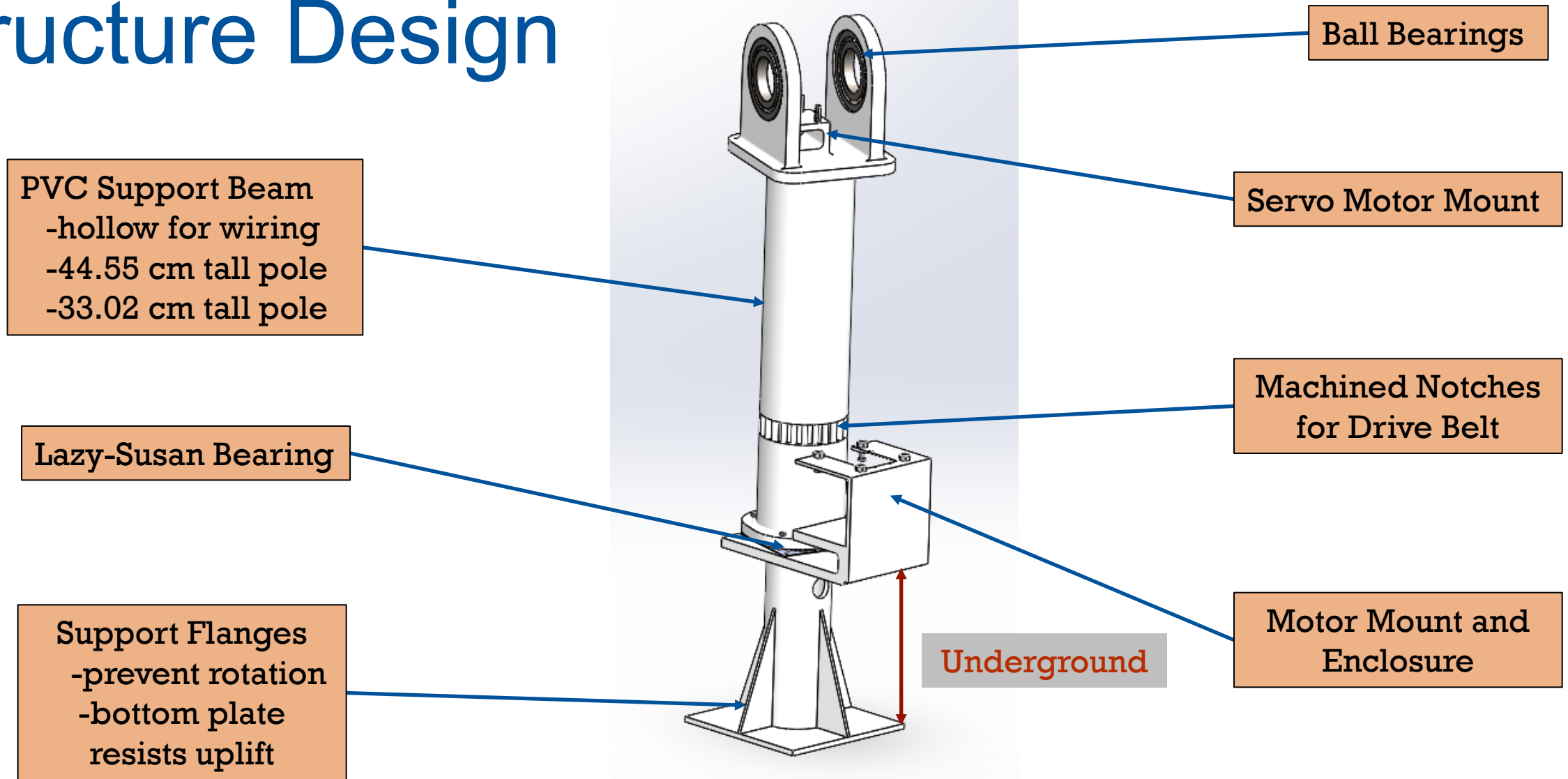
Motors can support load



Structure Subsystem

Need	Metric	Solution
3	$4 \leq \text{No. Heliostats} \leq 16$	4 reflective surfaces
6	No shading from other heliostats	Minimum spacing = 1.07 m
8	Overall cost $\leq \$100/\text{m}^2$	Hollow PVC tubing
10	Total area ratio ≈ 1	Proportional structure
13	Factor of safety = 2	Set standard for analysis
14	Operational lifetime ≥ 20 years	20+ years with maintenance

Structure Design



Subsystem Design Analyses

- Structure - Bending stress on the poles

Typical wind force on each mirror:

$$F_w = \frac{1}{2} \rho v^2 A = \frac{1}{2} (1.2 \text{ kg/m}^2) (4.47 \text{ m/s})^2 (0.25 \text{ m}^2) = 3.0 \text{ N}$$

Bending moment (per pole):

$$M = Fa = (6 \text{ N})(0.446 \text{ m}) = 2.67 \text{ N} \cdot \text{m}$$

Structure Subsystem Design Analyses

Second moment of area:

$$I = \frac{\pi}{4} (r_2^4 - r_1^4) = \frac{\pi}{4} [(50.8 \text{ mm})^4 - (38.1 \text{ mm})^4] = 3.58 \times 10^{-6} \text{ m}^4$$

$$\text{Bending stress: } \sigma = \frac{My}{I} = \frac{(2.67 \text{ N}\cdot\text{m})(0.0508 \text{ m})}{3.58 \times 10^{-6} \text{ m}^4} = 37.8 \text{ KPa}$$

Maximum bending stress at 90 mph: $\sigma = 3.07 \text{ MPa}$

Yield strength of PVC: 51.7 MPa

Structure Subsystem Design Analyses

- Torsional shear stress due to wind

$$\tau = \frac{Tc}{J}, \text{ where } J = \frac{\pi(d_o^4 - d_i^4)}{64}$$

$$\tau = \frac{(2.127 \text{ N} \cdot \text{m})(0.0508 \text{ m})}{\frac{\pi(0.1016^4 - 0.0762^4)}{64} \text{ m}^4} = 30.35 \text{ KPa}$$

Shear strength of PVC: 5 MPa

Structure Subsystem Design Analyses

- Maximum deflection at wind speed of 90 mph

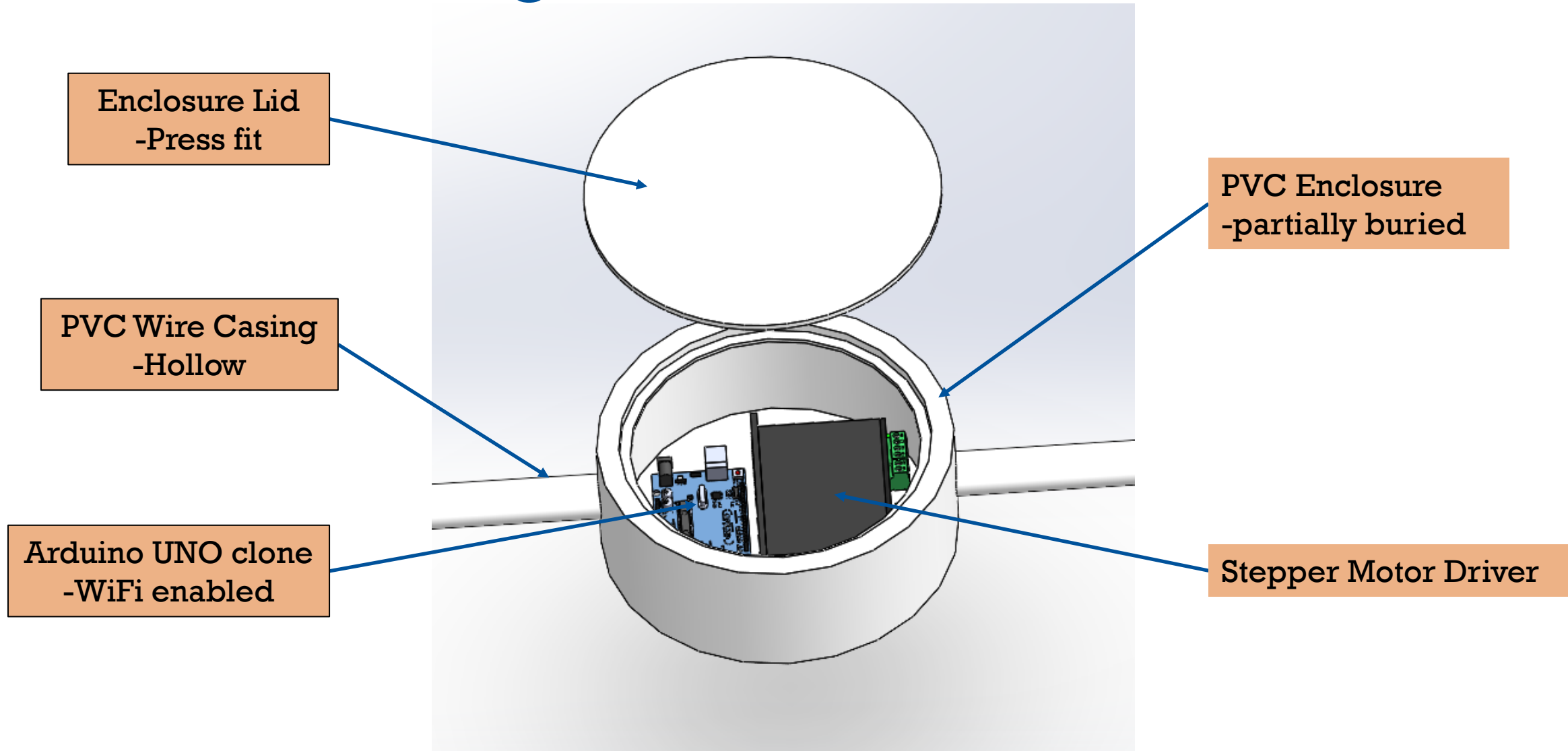
$$\delta = \frac{FL^3}{3EI}$$

$$\delta = \frac{(480 \text{ N})(0.4445\text{m})^3}{3(2.8 \times 10^9 \text{ m})(3.58 \times 10^{-6} \text{ m}^4)} = 1.403 \text{ mm}$$

Controller Subsystem

Need	Metric	Solution
5	Minimum refresh rate = 13.9 Hz	1 kHz refresh rate
8	Overall cost \leq \$100/m ²	Comparable Arduino UNO clone
9	Automated and computer controlled	WiFi capabilities
14	Operational lifetime \geq 20 years	20+ years with maintenance

Controller Design



Controller Subsystem Design Analyses

Controller & Driver – Heat Load (H) in Enclosure

$$h_{in} = \frac{P}{10} = \frac{194 \text{ W}}{10} = 19.4 \text{ W}$$

$$\Delta T = T_{max \text{ in Vegas}} - T_{target} = 46 - 40 = 6^\circ \text{C}.$$

This temperature difference correlates to a constant in W/m^2 (Table 1).

$$H = (SA_{box} * 11.3 \text{ W}/\text{m}^2) + h_{in} = (0.055918 \text{ m}^2 * 11.3) + 0.14 = \boxed{0.771873 \text{ W}}$$

Temperature Difference in Deg F	BTU/hr./sq. ft.	Temperature Difference in Deg C	Watts/sq.m
5	1.5	3	5.2
10	3.3	6	11.3
15	5.1	9	17.6
20	7.1	12	24.4

Controller Subsystem Design Analyses

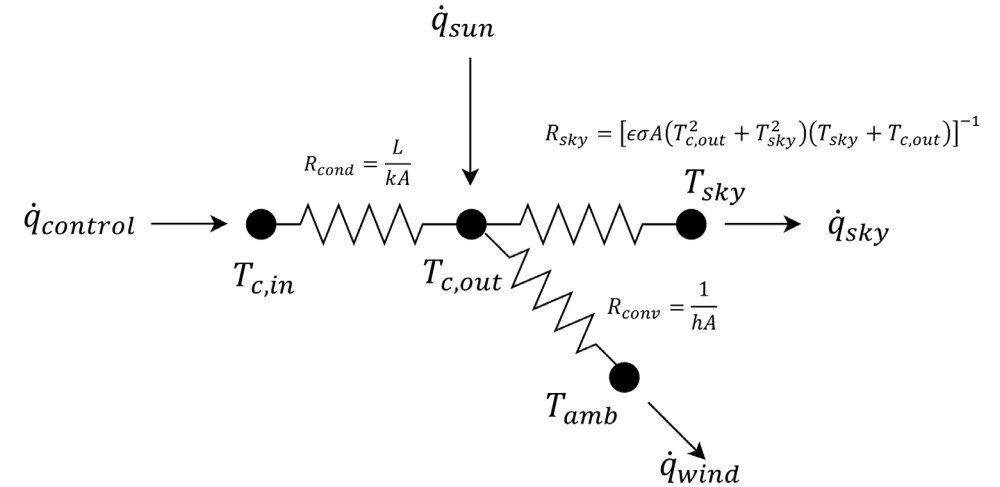
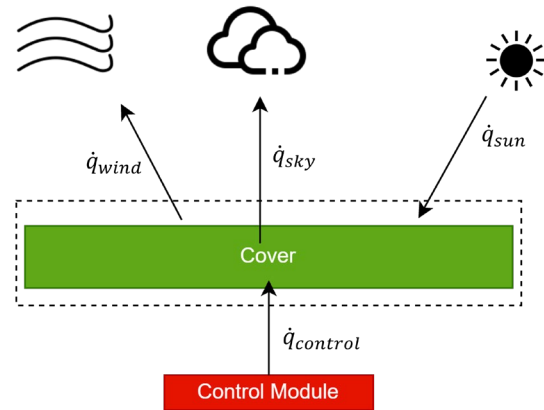
Max Enclosure Operating Temperature

$$\dot{q}_{control} = 19.4 \text{ W}$$

$$\dot{q}_{sun} = 32.45 \text{ W}$$

$$L = 0.00397 \text{ m}^2$$

$$A = 0.0324 \text{ m}^2$$



$$h = h_{wind} = 18.02 \frac{\text{W}}{\text{m}^2\text{K}}$$

$$k = k_{cover} = 0.2 \frac{\text{W}}{\text{mK}}$$

$$T_{amb} = 320.37 \text{ K}$$

$$T_{sky} = 318.10 \text{ K}$$

$$\sum \dot{q} = \dot{q}_{sun} + \dot{q}_{control} - \dot{q}_{sky} - \dot{q}_{wind} = 0$$

$$\dot{q}_{sky} = \frac{T_{c,out} - T_{sky}}{R_{sky}}$$

$$\dot{q}_{wind} = \frac{T_{c,out} - T_{amb}}{R_{conv}}$$

$$T_{c,out} = 379.23 \text{ K}$$

$$\dot{q}_{control} = \frac{T_{c,in} - T_{c,out}}{R_{cond}}$$

$$T_{c,in} = 391.06 \text{ K} = 117.91^\circ\text{C}$$

Cost Breakdown

Expense	Prototype Cost	Mass Production Cost
OTS Parts	\$139.97	\$97.98
Raw Materials	\$52.35	\$36.65
Manufacturing	\$19.26	\$16.05
Assembly Labor	\$4.20	\$4.20
Energy Consumption	\$0.67	\$0.67
TOTAL:	\$216.45	\$155.55

Full-scale plant production is expected to result in a 30% cost savings for bulk purchases.

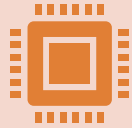
Summary



Thank you for your interest in HelioSmart



**WE STRIVE TO MEET EVERY
CUSTOMER NEED TO THE
BEST OF OUR ABILITY.**



**WE PROVIDE A COMPACT,
INNOVATIVE MULTI-UNIT
MODULE DESIGN.**



**UTILIZES EFFECTIVE COST
REDUCTION TECHNIQUES
WHILE PROVIDING
EFFECTIVE OPTICAL
EFFICIENCY AND
TRACKING ACCURACY.**



**PLEASE CONTACT US WITH
ANY ADDITIONAL
QUESTIONS OR CONCERNS.**