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# Sun Chariot

Innovative Heliostat Module Design

EML4501 Spring 2022 Section 29054

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#### **Presentation Outline**



#### 0

Hedgehog Concept



#### **Product Overview**

Subsystem Identification and Key Features

Engineering Analyses in Each Subsystem



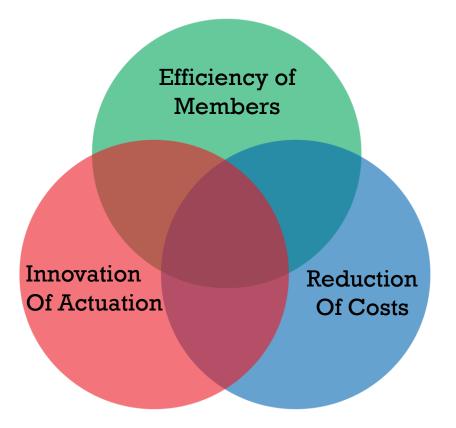
**Cost Breakdown** 



Conclusion

# Hedgehog Concept

 Sun Chariot is a heliostat design marked by its novel actuation which maximizes mechanical efficiency.





#### Identification of Subsystems



Reflector

#### Rotating Mechanism

Support Structure

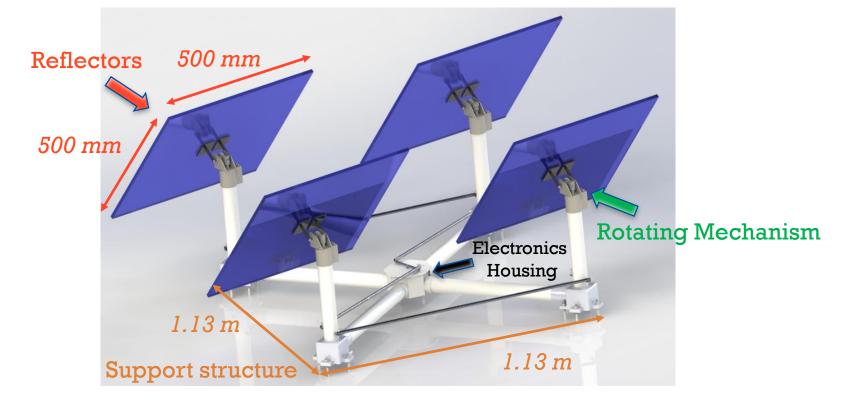
Electronics

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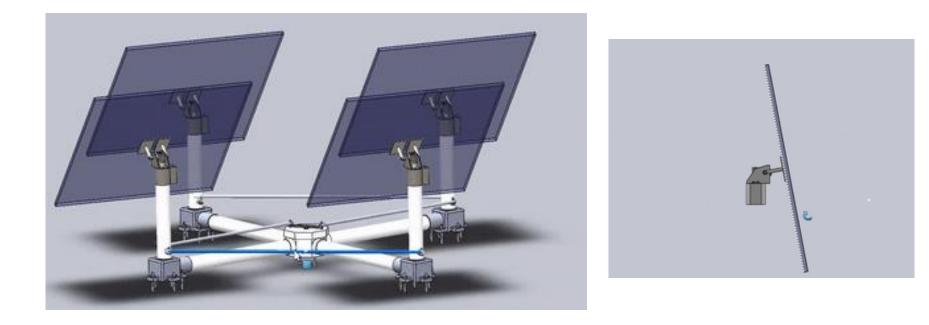
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#### **Product Overview**



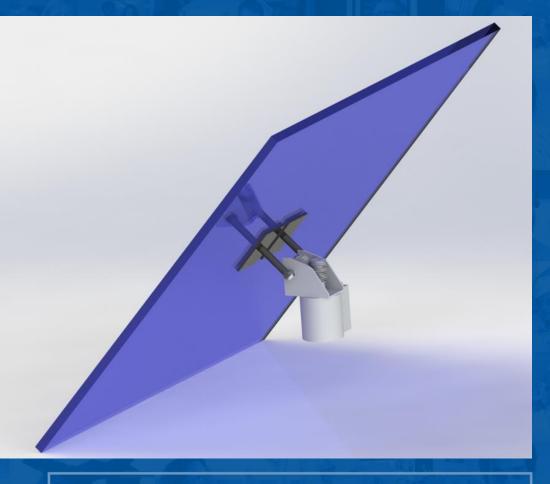


#### **Two-Axis Rotation**





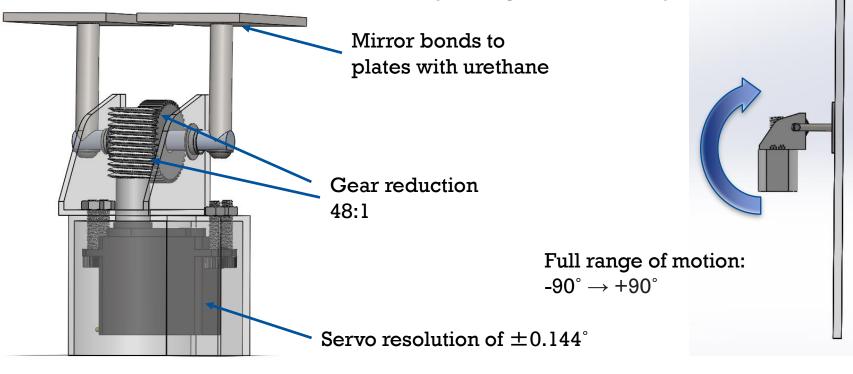
# Reflector



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#### **Degrees of Rotation (Major Axis)**



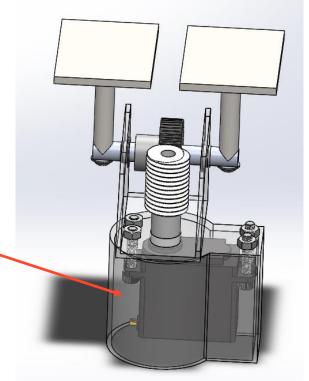
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# Degrees of Rotation (Major Axis)

Major axis motor: MS-R-6-40 Analog Servo

5 kg·cm max torque 4.8 Volts 360° rotation motor

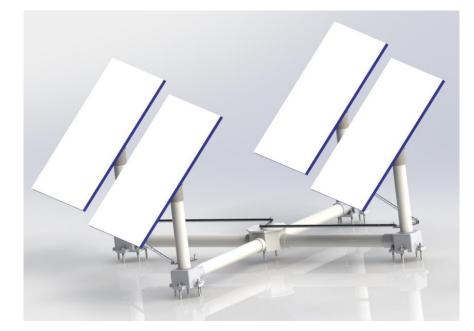


Torque after gear reduction: 7.01 N·m



#### **Reflector Sub-Assembly**

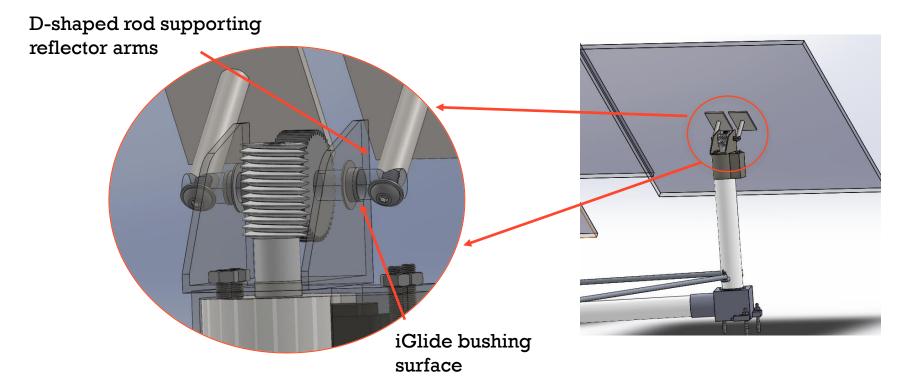
- Four Independently Moving Square Reflectors
  - 500 mm × 500 mm
- Silver-Coated Glass
- Fastened via Urethane Adhesive



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### **Simple Pivot Mechanism**





#### **Solar Concentration Ratio**

$$\frac{N_{reflectors} * A_{reflector}}{A_{reciever}} = SCR \qquad \qquad \frac{4250 * (0.25m^2)}{1 m^2} = 1063 suns$$

#### **Thermal Input Power**

 $\eta_{heliostat} * A_{heliostat} * N_{heliostats} * \dot{Q}_{sun} = P_{thermal input}$ 

0.98 \* 0.25 \* 4250 \* 1000 = 1.04 MW

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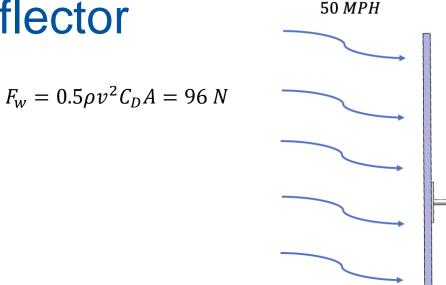
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#### **Thermal Fatigue**

 $\dot{O} = h(T_{max} - T_e)$ Glass: **Reflective Silver Mirror:**  $T_{max} = T_e + \frac{\dot{Q}}{h} = 49^{o}C + \frac{1000\frac{W}{m^2}}{10\frac{W}{m^2 \circ C}} = 149^{o}C$  $T_{max} = T_e + \frac{\dot{Q}}{h} = 49^{\circ}C + \frac{1000\frac{W}{m^2}}{24\frac{W}{m^2 0.00}} = 90.6^{\circ}C$  $T_{min} = -6.67^{\circ}C$  (Coldest temperature in placement location)  $\Delta T = 149^{\circ}C - (-6.67^{\circ}C) = 155.7^{\circ}C$  $\Delta T = 107^{o}C - (-6.67^{o}C) = 97.3^{o}C$ Otiaba, K. C., Bhatti, R. S., Ekere, N. N., Mallik, S., and Ekpu, M., "Finite element analysis of Hasselman, D. P. H., Badaliance, R., McKinney, K. R., & Kim, C. H. (n.d.). Failure the effect of silver content for Sn-Aq-cu alloy compositions on thermal cycling reliability of prediction of the thermal fatigue resistance of a glass - Journal of Materials Science. solder die attach," Engineering Failure Analysis Available: SpringerLink. Retrieved April 13, 2022, from https://www.sciencedirect.com/science/article/pii/S1350630712002166. https://link.springer.com/article/10.1007/BF00540926

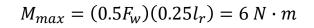
# Drag Force on Reflector

- $C_D = \text{Drag coefficient} = 1.28$
- A =Frontal area = 0.25  $m^2$
- $\rho = \text{Density} = 1.202 \frac{kg}{m^3}$
- $v = \text{Wind speed} = 50 \text{ MPH} (22.4 \frac{m}{s})$
- $F_w$  = Drag force on reflector



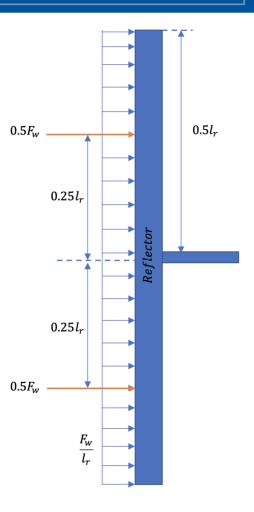
# **Reflector Bending Stress**

- $F_w$  = Reflector drag force = 96 N
- $l_r = \text{Reflector length} = 0.5 m$
- $w_r$  = Reflector width = 0.5 m
- $t_r$  = Reflector thickness = 0.005 m
- $\sigma_r$  = Bending stress
- n = Factor of safety



$$I = \frac{w_r t_r^3}{12} = 5.21 \times 10^{-9} m^4$$

$$y = \frac{t_r}{2} = 0.0025 m$$
$$\sigma_r = \frac{M_{max}y}{I} = 2.88 MPa$$
$$n = \frac{S_u}{\sigma_r} = 24.31$$





#### PP2401 Autourethane Strength

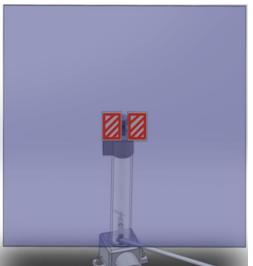
- $F_w$  = maximum wind force = 96 N
- w = width of application = 40 mm
- l = length of application = 60 mm
- $\sigma_y$  = strength of Urethane = 10.14 MPa
- $A_c$  = contact area
- $\sigma$  = maximum stress
- n = factor of safety

$$A_c = 2(w * l) = 4800 \ mm^2$$

$$\sigma = \frac{F_w}{A_c} = 0.02 \ MPa$$

$$n = \frac{\sigma_y}{\sigma} = 507$$

 \*\* High factor of safety due to large application area of Urethane (negligible effect on cost)





#### Customer Needs: Reflector Subsystem

Customer Need	Design Specifications	
Total reflective area $\leq 1 \text{ m}^2$	Total reflective area: 1 m <sup>2</sup>	$\checkmark$
Cleaning time $\leq 15$ minutes	Total cleaning time: 10 min	$\checkmark$
Focal thermal input power of at least 1 MW	Focal thermal input power: 1.04 MW	
Solar concentration ratio > 1000 suns	Solar concentration ratio: 1063 suns	<b>V</b>
Reflection distance $\geq 100 \text{ m}$	Reflective distance 100 m	



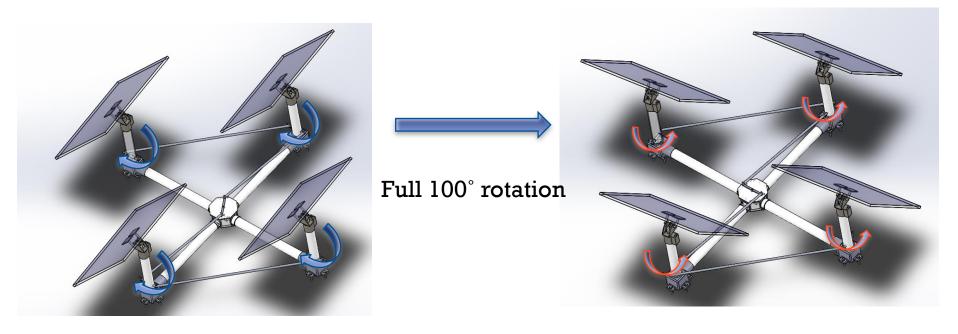
# Rotating Mechanism



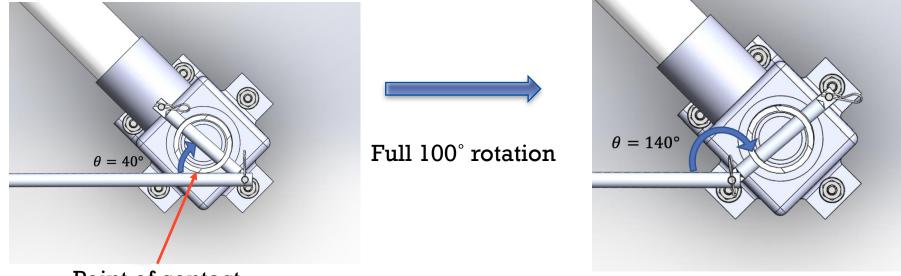
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#### **Minor Axis Rotation**



### **Minor Axis Rotation**



Point of contact

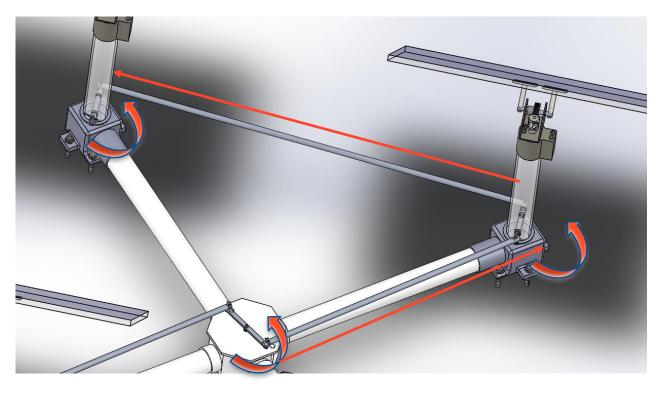
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#### **Rotating Mechanism**

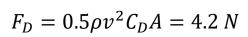
Central Motor: CYSS0090 Analog Servo Motor

10 kg ·cm Max Torque 4.8 Volts

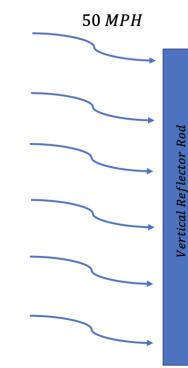


# Drag Force on Vertical Reflector Rod

- $C_D$  = Drag coefficient = 1.1
- $d_r = \text{Rod diameter} = 0.042 m$
- $h_r = \text{Rod height} = 0.325 m$
- $v = \text{Wind speed} = 50 \text{ MPH} \left(22.4 \frac{m}{s}\right)$
- $\rho = \text{Air density} = 1.202 \frac{kg}{m^3}$
- $F_D$  = Drag force on rod
- A =Frontal area  $= 0.25 m^2$



 $A = d_r h_r$ 





# **Vertical Rotating Rod Stress**

- $h_r$  = Height of rod = 0.325 m
- $d_r$  = Outer diameter of rod = 0.042 m
- $r_o$  = Outer radius of rod = 0.021 m
- $\rho$  = Density of air =  $1.202 \frac{kg}{m^3}$
- $C_{Dr}$  = Reflector drag coefficient = 1.28
- $C_{Dv}$  = Vertical rod drag coefficient = 1.1
- I = Moment of inertia = 7.9  $\times$  10<sup>-8</sup>  $m^4$
- v = Velocity of air
- F<sub>w</sub> = Drag force on reflector
- F<sub>D</sub> = Drag force on rod
- M<sub>r</sub> = Reaction moment

$$F_w = 0.5\rho C_{Dr} v^2 A_{ref} = 96 N$$

$$F_D = 0.5 \rho C_{Dv} v^2 h_r d_r = 4.2 N$$

$$M_R = h_r (F_w + 0.5F_D) = 31.88 \text{ N} * \text{m}$$

$$\sigma_{bending} = \frac{M_R r_o}{I} = 8.48 MPa$$

$$n = \frac{55 MPa}{8.48 MPa} = 6.49$$

$$F_w$$
  
 $h_r$   $F_D$   
 $0.5h_r$   
 $M_R$ 



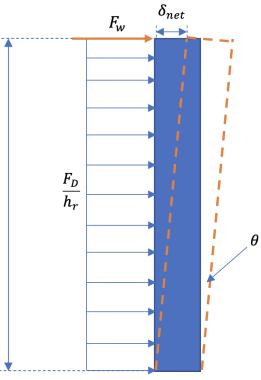
### **Vertical Rotating Rod Deflection**

- *E* = Modulus of elasticity of PVC = 4.425 *GPa*
- $\delta_w$  = Deflection due to reflector
- $\delta_D$  = Deflection due to distributed load
- $\theta$  = Angle of deflection



$$\delta_D = \frac{\left(\frac{F_D}{h_r}\right)h_r^4}{8EI} = \frac{F_D h_r^3}{8EI}$$

$$\delta_{net} = \delta_w + \delta_D = \frac{h_r^3}{EI} \left( \frac{F_w}{3} + \frac{F_D}{8} \right) = 3.2 \ mm \qquad h_r$$
$$\theta = \tan^{-1} \frac{\delta_{net}}{h_r} = 0.49^\circ$$

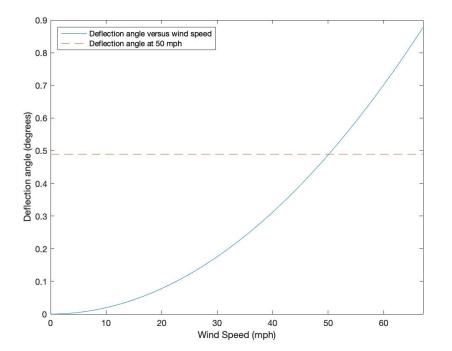




# Vertical Rotating Rod Deflection (Cont.)

 $\theta_{max} = 0.29^{\circ}$ 

<b>Rod Deflection at Various Speeds</b>				
Wind Speed (mph)	$\delta_{net} (mm)$	$\theta$ (degrees)		
10	0.13	0.02		
20	0.51	0.08		
30	1.15	0.18		
38	1.87	0.29		
50	3.20	0.49		



# **Resolution of Motor**

#### Reflector Servo

- $PW_{range}$  = pulse width range =  $2500\mu s$
- $\theta_{tot} = \text{total rotation} = 360^{\circ}$
- $W_{db} = \text{dead band width} = \pm 1 \mu s$
- *R* = resolution

$$\frac{PW_{range}}{\theta_{tot}} = 6.944 \mu s * degree^{-1}$$

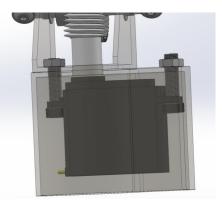
$$R = \frac{W_{db}}{6.944 \mu s * degree^{-1}} = \pm 0.144^{\circ}$$

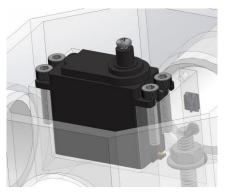
#### Central Servo

- $PW_{range} =$  pulse width range  $= 2200 \mu s$
- $\theta_{tot} = \text{total rotation} = 90^{\circ}$
- $W_{db} = \text{dead band width} = \pm 2\mu s$
- *R* = resolution

$$\frac{PW_{range}}{\theta_{tot}} = 24.444 \mu s * degree^{-1}$$

$$R = \frac{W_{db}}{24.444\mu s * degree^{-1}} = \pm 0.082^{\circ}$$







### **Reflector Servo Torque**

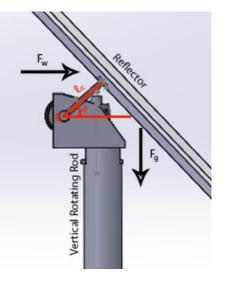
- $F_w = maximum wind force = 96 N$ •  $T_{stall} = stall torque = 0.2157 N \cdot m$
- $l_c = length \ of \ reflector \ support \ arm = 0.0412 \ m$   $T_o = output \ torque$
- $m_r = mass \ of \ reflector = 1.35 \ kg$
- $\eta_w = worm \ gear \ efficiency = 0.64$

- $\theta_r = angle \ of \ rotation$
- $T_c = torque applied to pinion rod$

•  $G = gear \ ratio = 48$ 



# **Reflector Servo Torque**

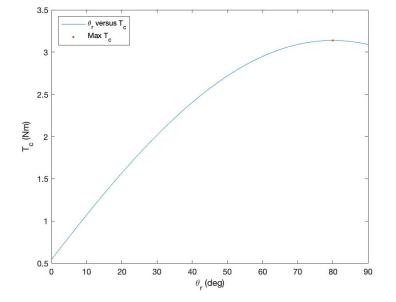


$$F_g = m_r g = 13.42 N$$

$$T_c = F_w l_c \sin\theta_r + F_g l_c \cos\theta_r$$

Peak at  $\theta_r = 79.9^{\circ}$ 

$$T_c = 4.0 N \cdot m$$





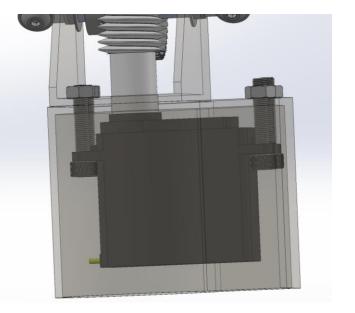
# **Reflector Servo Torque**

$$m_w = \frac{T_o}{T_{stall}}$$

$$m_w = \eta_w G$$

$$T_o = T_{stall} \eta_w G = 6.6 N \cdot m$$

 $T_o > T_c$ 





#### **Customer Needs: Rotating Mechanism**

Customer Need	Design Specifications	
Reflect light toward 100 m tall receiver	Reflector can redirect light to >100 m height	$\checkmark$
Total module cost $\leq$ \$100/m <sup>2</sup>	Total module cost \$301.06	X
Automatic rotation of 180° laterally and 90° longitudinally	180° rotation in major axis 90° rotation in minor axis	$\checkmark$
At least 10% cheaper than OTS	OTS parts used wherever possible	$\checkmark$
Each heliostat individually rotates 180°	Individual rotation of > 180°	$\checkmark$
Lifetime $\geq$ 20 years	Lifetime of 20+ years	$\checkmark$



# Support Structure



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#### Support Structure

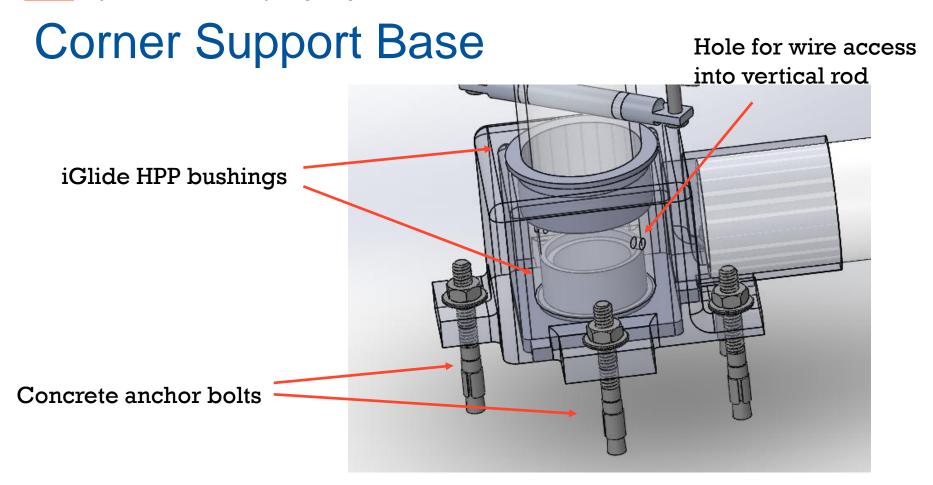
Four corner mounts

**PVC** Pipes

Central motor and Electronics housing Herbert Wertheim College of Engineering

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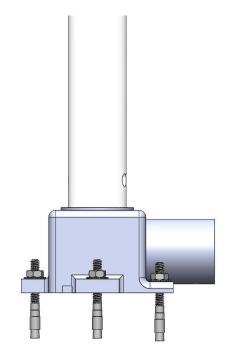




### **Base Flange Stress**

- F<sub>w</sub> = Drag force on reflector = 96 N
- $F_D$  = Drag force on rod = 4.2 N
- $h_r$  = Height of rod = 0.325 m
- $w_b = Width of base = 0.07071 m$
- $l_f$  = Length of flange = 0.01588 m

- $w_f$  = Width of flange = 0.0317 m
- t<sub>f</sub> = Thickness of flange = 0.01 m
- M<sub>A</sub> = Moment about point A (see fig.)
- $\sigma_f$  = Flange bending stress
- $\sigma_y$  = PETG Yield stress





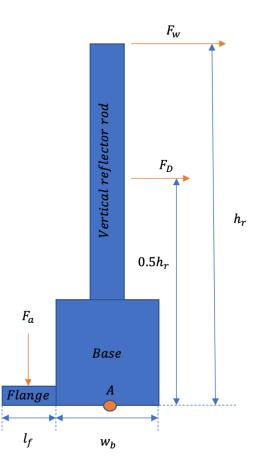
#### Base Flanges Stress Cont.

$$\sum M_A = F_a (0.5w_b + 0.5l_f) - F_w h_r - F_D (0.5h_r) = 0$$

$$F_a = \frac{h_r(F_w + 0.5F_D)}{0.5(w_b + l_f)} = 736.4 N$$

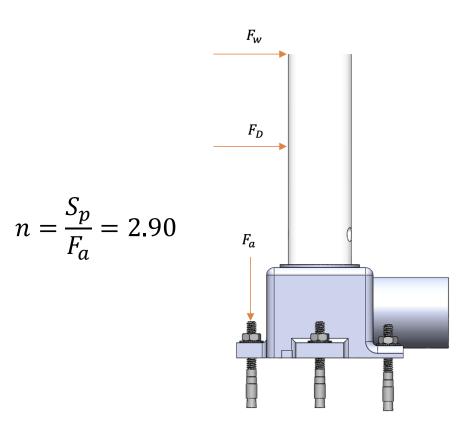
$$M = 0.5F_a l_f = 5.85 N * m$$

$$\sigma_f = \frac{My}{I} = \frac{My}{\frac{w_f t_f^3}{12}} = 11.7 \text{ MPa}$$
$$n = \frac{\sigma_y}{\sigma_f} = 4.27$$



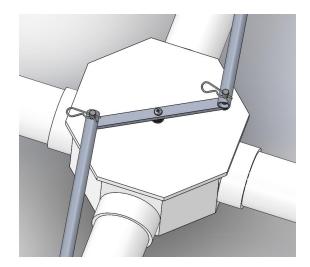
# Anchor Strength

- From flange analysis, F<sub>a</sub> = 736.4 N
- Pull-out strength of S<sub>p</sub> = 2135.15 N
- Safety Factor of 2.90, assuming load on one anchor

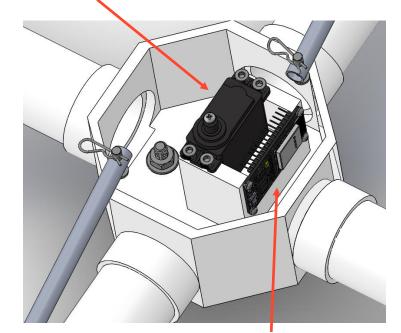




### **Middle Junction**



Central Servo Motor



Microcontroller

#### **Customer Needs: Support Structure**

Customer Need	Design Specifications	
Heliostat area $\leq 0.25 \text{ m}^2$	Single heliostat area 0.25 m <sup>2</sup>	$\checkmark$
Heliostats per module = 4 to 16	4 heliostats per module	$\checkmark$
Shading acceptable 60 min after dawn and 60 min before dusk	Designed to minimize shading within constraints	
Total module area $\leq 2 \text{ m}^2$	Total module area 1.35 m <sup>2</sup>	✓
Factor of safety $N \ge 2$	All factors of safety >> 2	$\checkmark$
Withstand weather conditions to at least 25% waterproofing	All materials water resistant and passed thermal analyses	$\checkmark$

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# Controls & Electronics

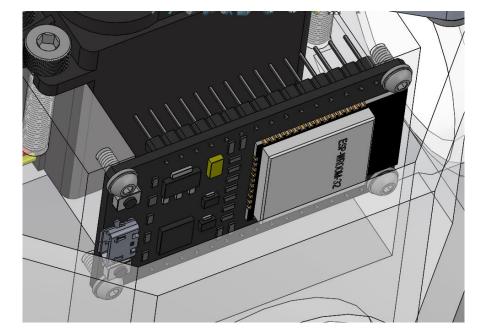


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### **Controls and Electronics**

- ESP8266 Wi-Fi enabled microcontroller
- 5.5 Volt power supply
- Safe Operating Temp. Range: -40 °C to 125 °C





### **Microcontroller Thermal Analysis**

- $k = \text{thermal conductivity constant} = 0.2 \frac{W}{mK}$
- $h = \text{convective coefficient of air} = 10 \frac{W}{m^{2}K}$
- $\sigma = \text{Stefan Boltzmann constant} = 5.67 \times 10^{-8} \frac{W}{m^2 K^4}$
- $\epsilon$  = emissivity = 0.84
- $\Delta x = \text{total length in contact with surface} = 48 \, mm$

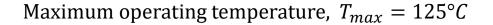
- $T_{surr}$  = ambient air temperature = 338.08 K
- $q_{cd}^{\prime\prime}$  = conductive heat flux
- $q_{cv}^{\prime\prime}$  = convective heat flux
- $q_r''$  = radiative heat flux
- $T_s$  = surface temperature

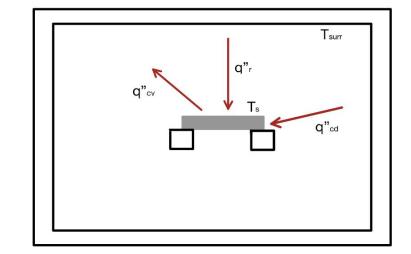


### **Microcontroller Thermal Analysis**

$$q_{cd}^{\prime\prime} + q_{cv}^{\prime\prime} + q_{r}^{\prime\prime} = 0$$
$$q_{cd}^{\prime\prime} = k \frac{\Delta T}{\Delta x}$$

 $q_{cv}'' = h(T_s - T_{surr})$  $q_r'' = \epsilon \sigma (T_s^4 - T_{surr}^4)$  $\therefore T_s = 357.251 \ K = 84.25 \ ^\circ C$ 





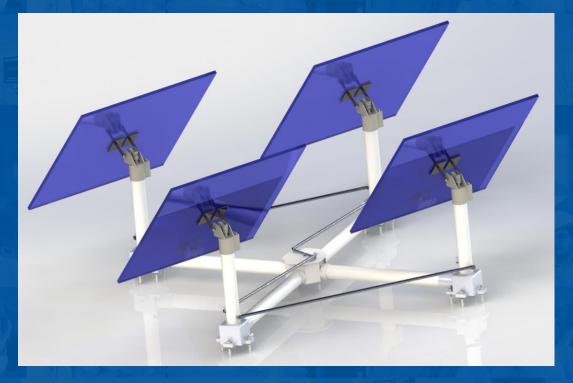


## Customer Needs: Controls & Electronics

Customer Need	Design Specifications	
Critical tracking error $\leq \pm 0.25^{\circ}$	Servo motor resolution of $\pm 0.144^{\circ}$	$\checkmark$
Active for 8.8 hours daily	Wifi enabled microcontroller can track at all hours of the day	$\checkmark$



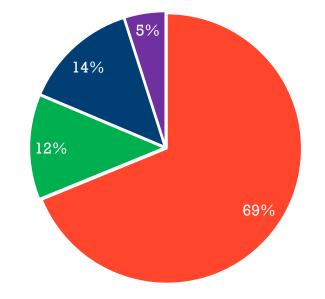
### Full Assembly



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### Cost Breakdown

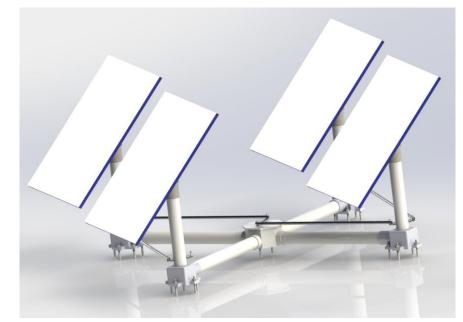
Category	Cost
Materials	\$205.83
3D Printing Components	\$37.47
Manufactured Parts	\$41.08
Installation	\$14.70
Total	\$301.06





### **Prototyping Defense**

- Modularity
- Low Number of Actuators
- Small Footprint Ratio
- Easily Sourced Materials
- Multipurpose Usage of Members



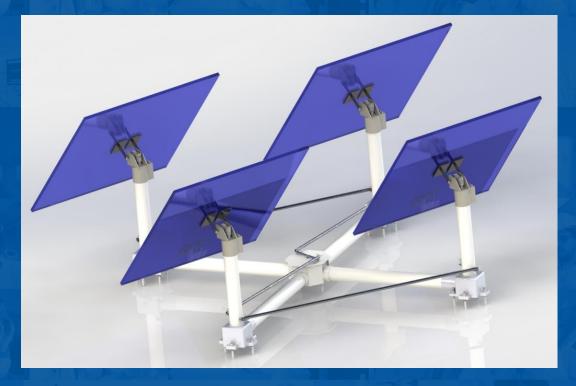


## Thank you for attending our presentation!





### Appendix

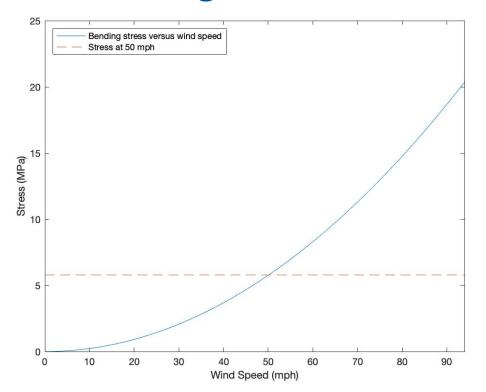


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### **Reflector Bending Stress Plot**



### Vertical Reflector Rod Wind Force

- $\rho = \text{density of air} = 1.202 \ kg/m^3$
- $\mu = \text{dynamic viscosity of air} = 1.825 \times 10^{-5} kg/m \cdot s$
- $h_r$  = height of rod = 0.325 m
- $d_r$  = diameter of rod = 0.042 m
- $v = velocity = 50 MPH = 22.4 \frac{m}{s}$

- A = frontal area
- $C_d$  = coefficient of drag
- $F_D = \text{drag force}$
- *Re* = Reynold's number

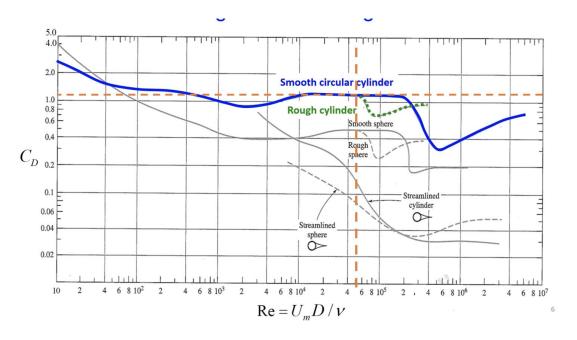


### Vertical Reflector Rod Drag Force

 $A = h_r d_r = 0.01365 \ m^2$ 

$$Re = \frac{\rho v d_r}{\mu} = 61687.25$$

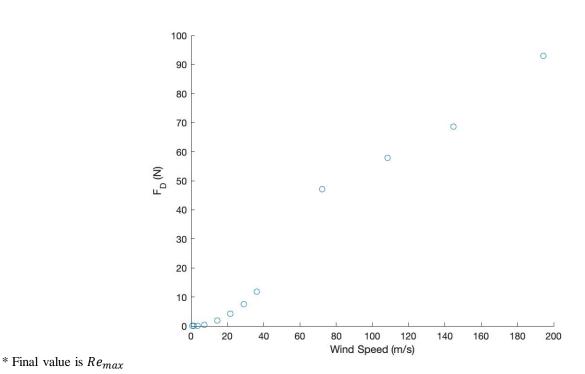
 $F_D = 0.5 \rho C_D v^2 A = 4.2 N$ 



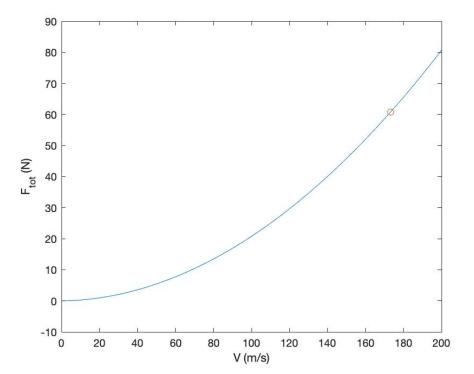


### **Vertical Reflector Rod Wind Force**

Selected $Re$ and $C_D$ pairs		
Re	C <sub>D</sub>	
$2 \times 10^{3}$	0.8	
$4 \times 10^{3}$	0.9	
104	1.1	
$2  imes 10^4$	1.1	
$4  imes 10^4$	1.1	
$6 imes 10^4$	1.1	
$8  imes 10^4$	1.1	
10 <sup>5</sup>	1.1	
$2 \times 10^{5}$	1.1	
3 × 10 <sup>5</sup>	0.6	
$4 \times 10^{5}$	0.4	
$5.3748 \times 10^5 *$	0.3	



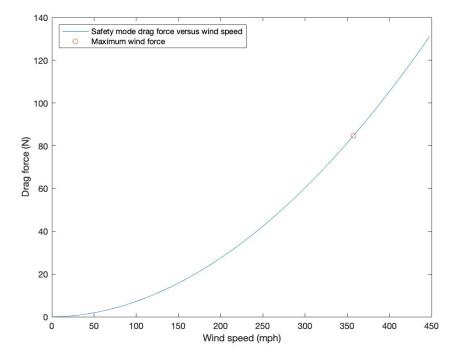
#### **Vertical Reflector Rod Wind Force**



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### Safety Mode Wind Speed



 $F_{w}$ 

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### Anchor Strength (Cont.)

 Anchor shear strength of S<sub>s</sub> = 1401.19 N

 $\Sigma F = 0 \rightarrow F_s = F_w + F_D$ 

$$F_s = 96 N + 4.2 N = 100.2 N$$

 Safety Factor of 13.98, assuming load on one anchor

